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PATENT

IN THE CLAIMS:

1. (Previously amended) A method of improving thermal comfort in a

passenger airplane, the airplane having a cabin with interior surfaces, the airplane

cabin for transporting one or more passengers, the method comprising:

applying a heat-reflecting coating with a low thermal emission coefficient to

at least one interior surface of a cabin of a passenger airplane,

whereby the coating provides improved radiation exchange with a passenger

in the airplane cabin.

2. (Original) The method of claim 1, wherein the coating is a

transparent conductive coating.

3. (Original) The method of claim 2, wherein the coating comprises a

conductive metal oxide.

4. (Original) The method of claim 2, wherein the coating comprising

indium tin oxide.

5. (Previously amended) The method of claim 1, further comprising

selecting a coating thickness to achieve a desired thermal emission coefficient for

the coating.

6. (Original) The method according to claim 5, wherein the thickness of

the coating, as applied, is less than 1 μ m.

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7. (Original) The method of claim 1, wherein the at least one interior

surface of the airplane cabin comprises at least one window of transparent plastic

material, and wherein the coating is applied to the at least one window.

8. (Original) The method of claim 7, wherein the at least one window

comprises polymethylmethacrylate

9. (Original) The method of claim 7, wherein the at least one window

comprises polycarbonate

10. (Original) The method of claim 1, wherein the at least one interior

surface of the airplane cabin comprises decorative plastic foil, and wherein the

coating is applied to the decorative plastic foil.

11. (Original) The method of claim 10, wherein the decorative plastic

foil comprises polyvinylfluoride.

12. (Original) The method of claim 10, wherein the decorative plastic

foil comprises polyvinylidenefluoride.

13. (Original) The method of claim 1, wherein the coating has a thermal

emission factor no greater than approximately 0.5.

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14. (Original) The method of claim 1, wherein the coating has a thermal emission factor no greater than 0.5.

- 15. (Original) The method of claim 1, wherein the coating has a thermal emission factor selected from the range of 0.1 to 0.3 inclusive.
- 16. (Original) The method of claim 1, wherein the at least one interior surface of the airplane cabin is associated with a lateral covering part, and wherein the coating is applied to the lateral covering part.
- 17. (Original) The method of claim 1, wherein the at least one interior surface of the airplane cabin comprises airplane glazing, and wherein the coating is applied to the airplane glazing.
- 18. (Previously amended) An airplane improved for thermal comfort, the improved airplane comprising:
 an airplane comprising an airplane cabin having interior surfaces,

a heat-reflecting coating with a low thermal emission coefficient on at least one of the interior surfaces,

whereby the coating provides improved radiation exchange with a passenger in the airplane cabin.

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19. (Original) The airplane of claim 18, wherein the coating is applied to

interior cabin walls.

20. (Previously amended) An airplane cabin part improved for thermal

comfort, the improved part comprising:

a part for use in an airplane cabin having at least one surface which, when

the part is installed in the airplane cabin, provides the at least one interior surfaces

of the airplane cabin,

a heat-reflecting coating with a low thermal emission coefficient applied to

the surface,

whereby the coated surface, when the part is installed in the airplane cabin,

provides improved radiation exchange with a passenger.

21. (Original) The improved part of claim 20, wherein the part is an

airplane window.

22. (Original) The improved part of claim 20, wherein the part is an

interior cabin wall panel.

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